

Flow Characterization of a Pulsating Heat Pipe through the Wavelet Analysis of Pressure Signals

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Abstract

Pulsating Heat Pipes are two phase passive heat transfer devices characterized by a thermally induced two phase oscillating flow. The correct detection of the dominant frequencies of such oscillations is fundamental to fully characterize the device thermofluidic operation but the studies available in the literature are very heterogenous and results are often discordant. In this work, the concept of dominant frequency in Pulsating Heat Pipes is thoroughly discussed and defined analytically. The wavelet transform is used to characterize the fluid pressure signal in the frequency domain varying the heat power input at the evaporator and in the condenser zone of a full-scale Pulsating Heat Pipe tested in microgravity conditions. During the slug-plug flow regime, the dominant frequencies falls in the range 0.6 – 0.9 Hz, showing an increasing trend with the heat load input. The Cross-Correlation reveals that the two signals at the evaporator and at the condenser are very similar. Finally, the instantaneous angle of phase is calculated and lies between 310 and 360 deg. This value can be physically interpreted as a repeatable time shift between the two signals that can be used to evaluate the flow local mean velocity (0.09 – 0.13 m/s) constituting a valuable alternative to the visualization techniques.

Keywords: Pulsating Heat Pipe, Frequency, Wavelet Cross Correlation, Flow velocity.

Nomenclature			
C_p	bending pressure loss coefficients	g	Gravitational acceleration (m/s^2)
ψ	complex mother function	f	Frequency (Hz)
C_ψ	admissibility condition	T	Temperature ($^{\circ}C$)
$\hat{\psi}(\omega)$	fourier transform	P	Pressure (Pa)
W_x	wavelet transform	N_{wt}	number of samples
a	scale dilation parameter (1/Hz)	N	Number of turns
τ	translation parameter (s)	\dot{Q}	heat load input (W)
P_{W_x}	wavelet power spectrum	c	damping action
W_{xy}	wavelet cross scalogram	m	Mass (kg)
$WLCC$	wavelet local correlation coefficient	ρ	Density (kg/m^3)
θ_{x-y}	instantaneous angle of phase (deg)	L	Length (m)
CoW_{xy}	wavelet co-scalogram	μ	dynamic viscosity (Pa s)
Subscripts			
w	wall	e	evaporator
D	dominant	c	condenser

Max	maximum	v	vapour
C	characteristic	l	liquid
Ac	acquisition	f	fluid
An	analysis		
Abbreviations			
PHP	Pulsating Heat Pipe	WT	Wavelet Transform
TS	ThermoSyphon	TC	Thermocouples
FT	Fourier Transform	ESA	European Space Agency
STFT	Short Time Fourier Transform	p	parabola
HHT	Hilbert-Huang Transform	FR	Filling ratio
TST	Time Strip Technique		

28 1 Introduction

29 Thermally induced oscillations in two phase confined slug flows largely affect the operation of
30 heat transfer devices such as micro-channel heat exchangers (MHE) and wickless heat pipes,
31 also known as Pulsating Heat Pipes (PHPs). Regarding MHEs, these phenomena may lead to
32 flow instabilities that are often detrimental for the device operation, causing a flow reversal to
33 the inlet manifold [1] while, in PHPs, they constitute their very basic working principle [2].
34 The frequency analysis of experimental data has been proved to be a promising tool to
35 characterize such unstable behaviours in particular through the detection of dominant
36 frequencies in the flow motion. The present work proposes to apply the wavelet transform to
37 the fluid local pressure signal of a PHP in the evaporator and in the condenser zones, to
38 rigorously define the concept of dominant and characteristic frequencies and link them to the
39 device physical behaviour. Frequency analysis on PHP have already been performed in the
40 literature but, as shown in the following state of the art, results are quite heterogeneous; Table
41 1 summarises, in chronological order, the most relevant works available in literature so far.
42 They can be distinguished according to three main factors: i) the definition given for the
43 dominant frequency of a PHP signal; ii) the type of time-frequency technique; iii) the type of
44 experimental signal exploited for the analysis.

45 [The obtained results show that the wavelet tool can detect the dominant frequencies in the](#)
46 [system and used to extrapolate the mean fluid local velocity only by means of pressure signals,](#)
47 [avoiding the complexity and cost of visualization techniques \(i.e. transparent inserts and high](#)
48 [speed cameras.](#)

Table 1. resume of the literature review

Authors name	Year	Geometry		I.D./O.D. [mm]	Working fluid	Signal	Section	Type of analysis	Frequency [Hz]	Vol. F.R. %	Heat input \dot{Q} [W]	
		Cross-section	Turn									
Jong-Soo et al. [3]	2003	rectangular	10	1.5x1.5	R-142b*	P_f	eva	FT	1.22;1.85 1.71;2	40 60	-	
Xu et al. [4]	2005	tubular	3	2/3	FC-72**	T_w	eva	FT	0.09	70	12	
									close to zero; 0.46; 0.9		25.6	
							-		12			
							close to zero; 0.46		25.6			
Khandekar et al. [5]	2009	tubular	1	2/4	ethanol	P_f	eva	FT	0.1 – 3	60	20	
Zhao et al. [6]	2011	tubular	6	1.65-3.18	water	T_w	eva	FT	-	49.2	210 - 240	
								WT	-			
Mameli et al. [7]	2012	tubular	2	2/4	ethanol	P_f	adia	FT	-	-	40 -100	
Peng et al. [8]	2013	tubular	8	4	ammonia	liquid slug velocity		HHT	$C_b = 0$	3.2	20	-
										2.5	30	
					2.4					40		
					2.3					50		
					2.7					60		
					2					70		
acetone	$C_b = 0.2$	1.8	60									
water	$C_b = 0.5$	1										
Mameli et al. [9]	2014	tubular	2	2/4	ethanol	P_f	adia	FT	-	65	40-100	
Fairley et al. [10]	2015	flat	6	1.5	water	T_w	eva	STFT energy spectrum	< 0.2	70	210 - 240	
									0.2-0.4		210; 230	

								HHT energy spectrum	0.1		240
								HHT spectrum	< 0.2		210 - 240
Karthikeyan et al. [11]	2015	Tubular	5	2.5/6	water CuO nanofluid	T_f	eva cond	FT	-	50	50
Spinato et al. [12]	2015	Flat	1	1	R245fa***	T_w	eva	TST	-	20	8
									0.7		22
									-		36
									1.2	60	6-8
									-		14-32
									1.2		34
Monroe et al. [13]	2017	tubular	4	3.25/4.8	water	T_f	eva adia cond	FT	~2	70	60
									~3		150
						~4			275		
						T_w			-		
Chi et al. [14]	2018	tubular	15	1/2	ethanol	T_w	eva	FT	0-0.1	6.7 - 40	14 20 25
Ishii et al. [15]	2019	Flat	12	-	water	T_f	eva	FT	-	50	74.4
									<3		118.3
											172
											238
									2		316
	405										
Takawale et al. [16]	2019	Tubular	17	1/2	ethanol	P_f	cond	FT	-	60	64-112

Legend: FT is Fast Fourier Transform. STFT is Short Time Fourier Transform. HHT is Hilbert-Huang Transform. TST is Time Strip Technique. WT is Wavelet Transform. P_f is fluid pressure. T_f is fluid temperature. T_w is wall temperature. C_b is bending pressure loss coefficients. * Chlorodifluoroethane ** Perfluorohexane. *** Pentafluoropropane.

51 *1.1 The dominant frequency*

52 Only a few authors provided a definition of dominant frequency so far in the literature. Xu et
53 al. [4] are among the first to introduce the time-frequency analysis as a tool for the PHPs
54 thermofluidic investigation and they define the dominant frequency as a peak of energy in the
55 Power Spectrum. Their work exploits the Fourier Transform (FT) analysis on three wall
56 temperatures signals, in the evaporator and the condenser section at different heat loads. At 12
57 W, the dominant frequency values (0.1 Hz) are much smaller than those obtained for 25.6 W
58 (0.46 – 0.9 Hz). Additionally, Karthikeyan et al. [11] applied the FT on the temperature signal
59 at different working fluid, deionized water and CuO nanofluid. The frequency wasn't found by
60 means the time-frequency technique, and the authors concluded that the results depend on the
61 chaotic nature of the PHP oscillation. In Jong-Soo et al. [3], the Fourier Transform is achieved
62 on the pressure signal and the dominant frequencies were determined to be 1.22 Hz and 1.85
63 Hz (at 40 vol.%) and 1.71 Hz and 2 Hz (at 60 vol.%). Mameli et al. [7][9] performed the FT
64 analysis on the pressure signal at different heat input levels without recognizing any peak in the
65 power spectra and thus opening the discussion on whether the FT would be the most suitable
66 tool for the frequency detection. Also, in Takawalea et al. [16], the FT is applied on the pressure
67 signals, but the dominant frequencies cannot be found too. Despite the FT analysis has been
68 exploited more extensively with respect to other techniques, it is intrinsically not able to detect
69 the time range where the dominant frequencies occur, so the analysis is often performed on very
70 large amount of data and it often provides noisy outputs where peaks are hardly recognizable.
71 In this vein, Fairley et al. [10] describe the presence of energy peaks in the Power Spectrum as
72 occurrence of intermittent high-energy oscillations in the PHP evaporator temperature signal.
73 In their study, the Short-Time Fourier Transform (STFT) and Hilbert-Huang Transform (HHT)
74 analyses are performed on the wall temperature and at different heat loads. The Hilbert-Huang
75 energy spectrum shows intermittent oscillations at frequencies between 0.2 and 0.4 Hz. In the
76 Hilbert-Huang marginal spectra, the intermittent high-energy peaks fall in the range 0.02-0.2
77 Hz. The power spectrum of temperature signal tends to decrease when the heat input increases,
78 in accordance with Ishii et al. [15] and Peng et al. [8] but in opposition with Spinato et al. [12]
79 and Monroe et al. [13]. The results of the STFT are like those obtained with HHT, but the HHT's
80 sharper time and frequency resolution makes some features of the energy spectra more evident.
81 Overall, works derive a definition of dominant frequency only from a visual interpretation of
82 graphical results.

83 In Spinato et al. [12], the Time Strip Technique (TST) [12], a novel synchronized thermal and
84 visual investigation technique, is used to assess the thermal and hydrodynamic behaviours in a
85 single loop PHP. The 3D frequency spectra are computed for the local evaporator and condenser
86 wall temperatures, as well as for the time-strip (or interval in the time domain) intensity at the
87 same locations. Dominant frequencies are 0.6 Hz for FR = 20% and 1,2 Hz for FR = 60%. The
88 condenser signal results are almost absent, compared to the result obtained at the evaporator. In
89 the present work, the Wavelet Transform (WT) is chosen because it can overcome the limits of
90 FT and STFT. In this sense, the WT is a powerful tool with a good capacity for time-frequency
91 localizations and multiresolution representations, giving the possibility to “locate” the eventual

92 dominant frequency in time [17] [18]. A rigorous and comparable way to define the dominant
93 frequency of a PHP starting from the wavelet analysis of the fluid pressure signal is proposed
94 in section 3.2.

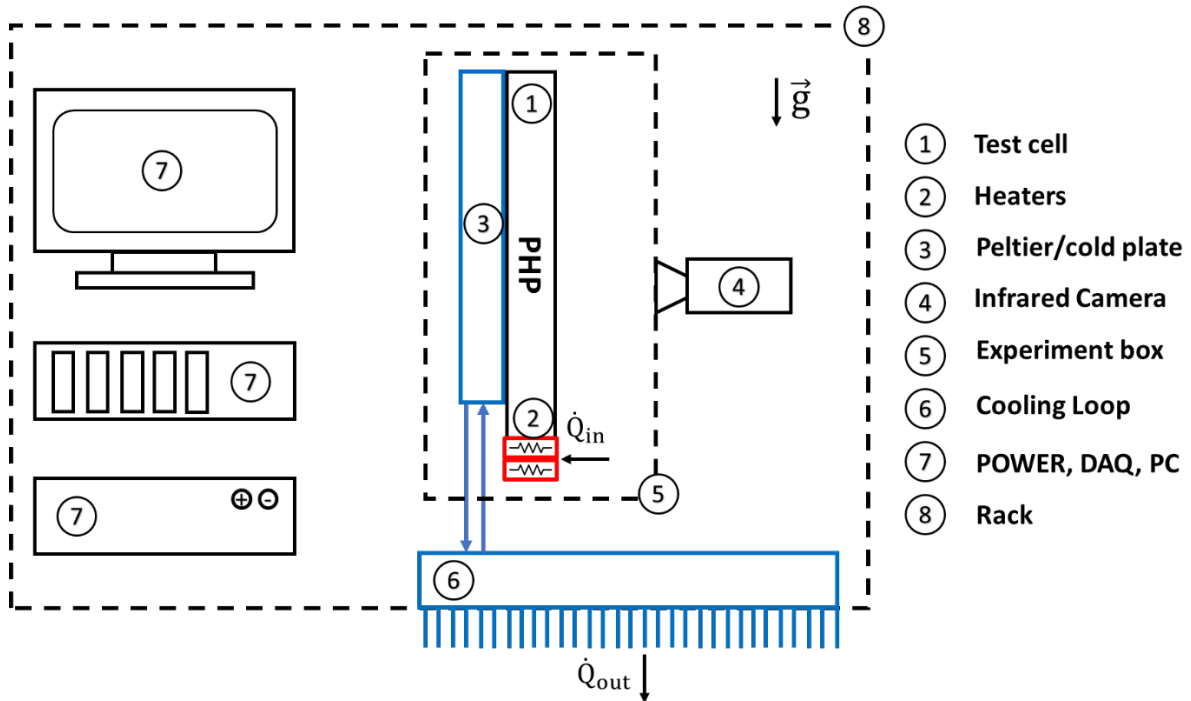
95 *1.2 Choice of the PHP signals*

96 All the techniques mentioned in the previous section are applied to different types of thermo-
97 fluidic signals. Much research analyses the PHP pseudo-steady state, and the most common
98 choice for the signal is the wall temperature. Zhao et al. [6] apply the FT and the WT to the
99 wall temperature signal but no dominant frequency is identified. Chi et al. [14] use the FT on
100 the wall temperature signal too and find frequencies up to 0.1 Hz. Monroe et al. [13] are among
101 the first that compare the results obtained by the wall temperature signal analysis with those
102 obtained by the fluid temperature signal analysis. In their work, two thermocouples (TCs) are
103 placed inside (fluid) and on top (wall) of an operating PHP. The results show that the
104 frequencies of the fluid temperature fall in the range from 2 to 4 Hz, while no frequencies are
105 detected by the wall temperature analysis, due to the thermal inertia of the tube. The FT is also
106 applied by Ishii et al. [15] on the fluid temperature signal. For heat loads of 74.4 to 112 W,
107 there is not any visible peak in the power spectrum, while for 316 W and 415 W the peaks are
108 identified around 2 and 3 Hz. In addition to the signals recorded during the experimental tests,
109 some authors apply the frequency analysis to the numerical output of their models but this may
110 be misleading if the model is not validated, at least partially, against experimental data. Peng at
111 al. [8], for example, present a nonlinear thermomechanical finite-element model of PHP that
112 aims at simulating the slug oscillation, and then calculate the time-varying spatial distributions
113 (HHT) of the slugs velocity by varying different parameters. If the bending pressure loss
114 coefficients C_b and the heat input increase, the frequencies decrease and they fall in the range
115 1-3.2 Hz. Finally, some authors, apply the Fourier Transform on the fluid pressure signal (e.g.
116 [7], [9]). Among these works Khandekar et al. [5] focus on a steady state and transition state
117 of operation PHP. The frequency falls in the range from 0.1 to 3 Hz. The choice of the fluid
118 pressure signal brings substantial advantages with respect to the others signals (wall and fluid
119 temperatures) because it is directly related to the fluid motion, it is not affected by the probe
120 thermal inertia, thus, it allows to acquire data at a very high sampling rates. As shown in the
121 next sections, the wavelet analysis presented here is performed at first on all the three signals
122 (wall temperature, fluid temperature and fluid pressure) to give proof of the above advantages,
123 then the fluid pressure signal is kept as reference for the whole post processing.

124 Beyond the standard WT frequency analysis of singular signals, the Cross-Correlation and the
125 calculation of the instantaneous angle of phase between the two signals at the evaporator and at
126 the condenser is also performed to demonstrate that the last can be physically interpreted as a
127 repeatable time shift between the two signals that can be used to evaluate the flow local mean
128 velocity. The flow velocities values obtained by means of the WT analysis are finally compared
129 to those obtained by the InfraRed visualization technique showing a good match and opening a
130 new and valuable alternative for the PHP flow characterization where visualization techniques
131 are not applicable.

132 **2 Experiments and post-processing tools**

133 **2.1 Experimental equipment in Flight**



134 Figure 1: Overall experimental layout.
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136 The experimental equipment in flight configuration is shown in Figure 1. It consists of the test
137 cell (1) with two electric heating elements (2) in the evaporator zone. The condenser zone
138 temperature is controlled by a Peltier cells array (3), coupled with a cold plate where distilled
139 water is recirculated and cooled by a secondary cooling loop (6) able to dissipate the whole
140 heating power. An infrared camera (4) allows to record the flow regimes and the fluid
141 temperature distribution through the transparent section made of sapphire. Signals are acquired
142 using a Data Acquisition system described in detail in the next sections (7). The subsystem
143 containing fluids are inserted in a double containment (5) and finally, all the systems described
144 above are mounted on a rack structure.

145 **2.2 Test cell**

146 The PHP test cell consists of an annealed aluminum tube with an inner/outer diameter of 3/5
147 mm and it is also thoroughly described by Mameli et al. [19]. The tube is folded in a single
148 loop staggered configuration with fourteen turns in the evaporator zone as shown in Figure 2.

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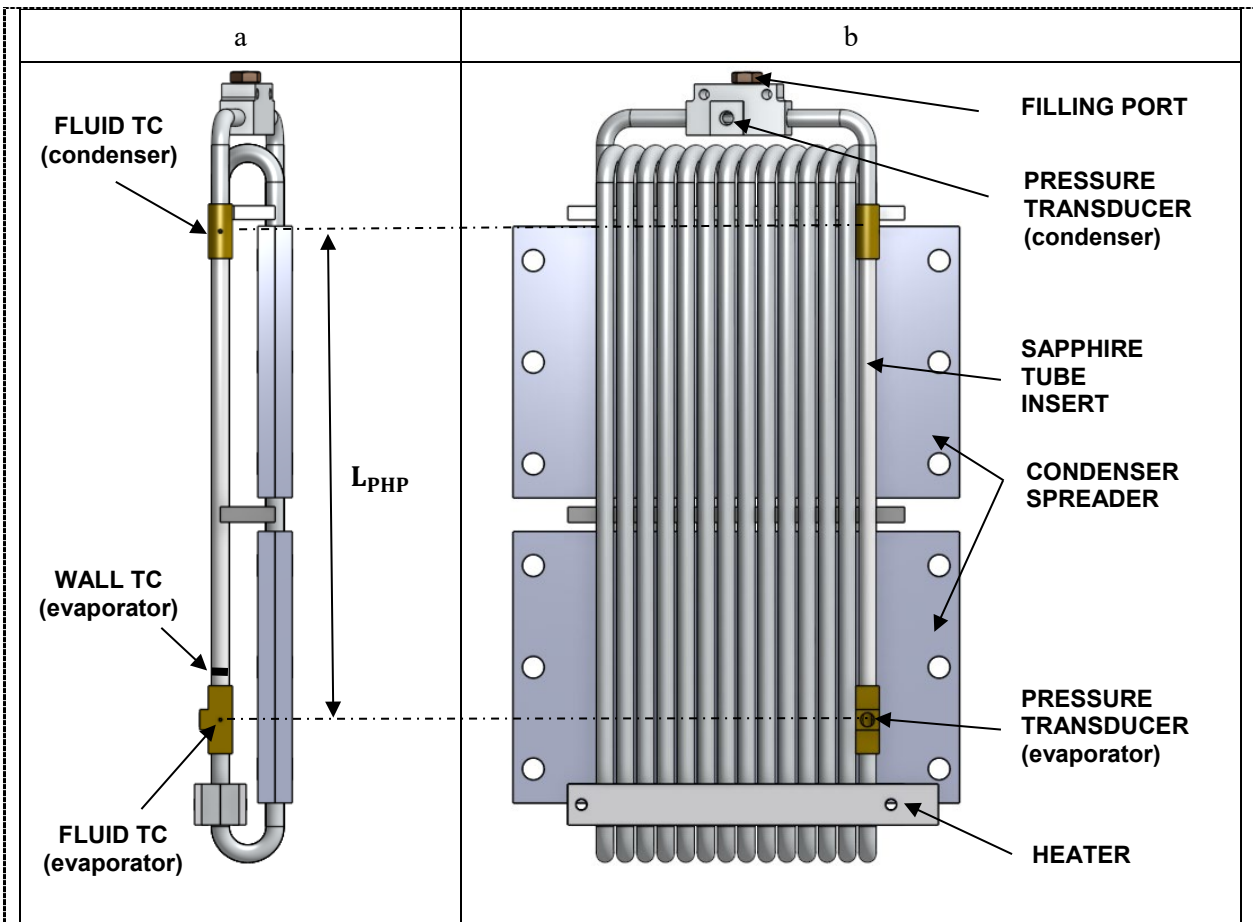


Figure 2 Test cell CAD (220x80x25 mm) side view (a) and front view (b) with pressure transducers and thermocouple locations of the fluid signals.

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An aluminium T-junction (on the top of figure 2 b) allows to close the loop and hosts a miniaturized pressure transducer along with the vacuum and filling port. Two brass connections allow to embed a sapphire tube insert, transparent (transmissivity 0.9) both to the visible and the Medium Wave InfraRed (MWIR), and to host two K-type micro-thermocouples for the fluid temperature measurement, as well as one pressure transducer close to the evaporator section. Two ceramic ohmic heaters supply from a minimum of 18W to a maximum of 180W by means of a programmable power supply, corresponding to an average wall to fluid heat flux from 1.10 to 11.43 W/cm². In particular, the tested heat input levels are $\dot{Q} = 68, 96, 134, 146 \text{ W}$. The backside of the condenser spreaders is cooled down by means of eight Peltier cells and cold plate temperature control system (not shown in figure 2). The main measured variables along with the probes specifications and their maximum errors are shown in Table 2. The device is partially filled with 22 ml of FC-72 (50% vol.). The test rig is then loaded on an Airbus A310 and a total of 93 parabolic trajectories (hypergravity-microgravity-hypergravity) are performed over the three days of flight. The device is oriented in bottom heated mode (the main acceleration field in the flow path direction). During the thermal characterization, the device is

174 heated up at the desired power level before the occurring of the microgravity period, and the
 175 power level is kept constant for the whole sequence of parabola. The pressure signals are
 176 acquired at 200Hz and the temperature signals as well as the MWIR camera are acquired at
 177 50Hz.

178 Table 2. Measured quantities and uncertainties.

Parameter	Specs.	Max. error
Tube wall temperature	T-Type thermocouples (0.5mm bead diameter)	± 0.1 °C
Fluid temperature	Omega® KMTSS-IM025E-150 K-Type thermocouples (0.25 mm bead diameter)	± 0.2 °C
Fluid Pressure	Keller® PAA-M5-HB, 1bar abs	± 500 Pa
Fluid temperature in sapphire section	AIM® MWIR camera, (wavelength 3-5 μ m)	± 2 °C
Power Input	GW-Instek®, PSH-6006A	± 3 W

179 2.3 Infrared Analysis

180 Since the sapphire tube is almost transparent to the radiation in the MWIR spectrum, the
 181 images, allowed to measure, not only the temperature distribution of liquid plugs, as already
 182 demonstrated in a previous work by the same authors [20], but also their length and velocity.
 183 The sapphire tube portion framed by the camera is 65.8 mm long and results in a 701x54 pixel
 184 image, therefore the pixel dimension is approximately 93 μ m. Figure 3 shows two consequent
 185 time frames of the fluid temperature distribution (orange line) in the mid axis of the transparent
 186 extracted from the MWIR images. Note that only when the liquid phase fills completely the
 187 channel (liquid slugs), the temperature values are detectable (continuous line). For all the other
 188 cases, i.e. vapor bubbles, or dispersed flow, since the vapor phase is almost transparent to the
 189 MWIR spectrum, the corresponding temperature is not physically meaningful (dashed line). In
 190 any case, the transition regions between higher and lower temperatures correspond to the
 191 menisci of a liquid slug. A thorough image analysis is performed by means of a custom Matlab
 192 software to detect the liquid slug menisci at each frame and calculate their length and velocity.
 193 The liquid slug recognition is performed evaluating the local maxima and minima of the
 194 temperature first derivative (black line): a liquid plug is always detected between a maximum
 195 and a minimum. Being an adiabatic zone, no evaporation or condensation occurs while the slugs
 196 travels through the sapphire tube, thus slugs are expected to maintain the same length
 197 throughout all the recorded frames. Indeed, the maximum discrepancy in the length
 198 measurement between different frames is 5 pixels and this occurs for less than the 20% of the
 199 frames. The *i*-th slug velocity over one time step, i.e. between frame *k-1* and frame *k*, is defined
 200 as

$$v_{i,k} = (x_{i,k} - x_{i,k-1}) f_{ac} \quad (1)$$

201 where $x_{i,k}$ is the i -th slug center point in frame k , $x_{i,k-1}$ is the i -th plug center point in frame
 202 $k-1$ and f_{ac} is the acquisition frequency. It follows that the maximum uncertainty of the
 203 instantaneous velocity is $0.093 \cdot 10^{-3} \cdot 5 \cdot f_{ac} \cong 0.023m/s$. The mean velocity is finally
 204 evaluated as

$$\bar{v}_{i,IR} = \frac{1}{N_f} \sum_{k=1}^{N_f} v_{i,k} \quad (2)$$

205 where N_f is the total number of frames. In this case, since the maximum discrepancy in the
 206 length measurement between different occurs for less than the 20% of the frames, the maximum
 207 uncertainty is tempered as follows $0.2 \cdot 0.093 \cdot 10^{-3} \cdot 5 \cdot f_{ac} \cong 0.004 m/s$. The mean
 208 velocity values are used in section 3.6 as a reference for the comparison with the mean velocities
 209 extracted through the WT analysis.

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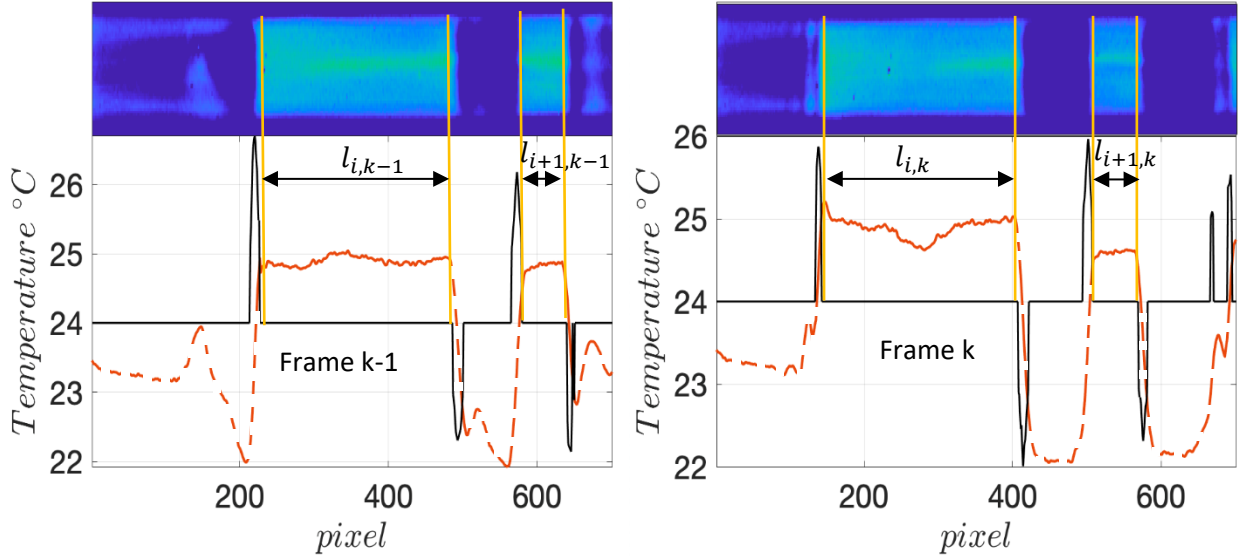
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221 Figure 3: On top, IR images of the sapphire tube. On the bottom, the temperature distribution (orange),
 222 the first derivative of temperature distribution (black), plugs ends (yellow).

223 2.4 Wavelet Transform methodology

224 The fluctuations of the pressure and temperature signals are characterized through techniques
 225 based on the continuous Wavelet Transform (WT). As described by Buresti et al. [17], a wavelet
 226 function can be any real or complex function $\psi(t) \in L^2$ that satisfies the following admissibility
 227 condition:

$$C_\psi = \int_{-\infty}^{+\infty} |\hat{\psi}(\omega)|^2 |\omega|^{-1} d\omega < \infty \quad (3)$$

228 where $\hat{\psi}(\omega)$ is the Fourier transform of $|\psi(t)|$. Indeed, to guarantee the reversibility of the
 229 transform, C_ψ must be a finite quantity and, in practice, this implies that $\psi(t)$ has zero mean

230 value. Under this admissibility condition, the Wavelet Transform $W_x(a, \tau)$ can be defined as
 231 follows:

$$W_x(a, \tau) = \frac{1}{\sqrt{a}} \int_{-\infty}^{+\infty} x(t) \psi^* \left(\frac{t - \tau}{a} \right) dt \quad (4)$$

232 where $a \in \mathfrak{R}^+$ is the scale dilation parameter and $\tau \in \mathfrak{R}$ is the translation parameter. In the
 233 present study, the complex Morlet wavelet $\psi(t) = e^{i\omega_0 t} e^{-t^2/2}$ is used, with a central frequency
 234 $\omega_0 = 2\pi$ in order to well-balance time and frequency resolutions (Table 3). For the Morlet
 235 wavelet the frequencies f are related to the scales a by $f = \omega_0/(2\pi a)$, thus in this case, $f =$
 236 $1/a$. The wavelet power spectrum can be obtained from the integration in time, i.e, for each
 237 scale/frequency, the wavelet energy map, as:

$$P_{W_x}(a) = \frac{1}{C_\psi} |W_x(a, \tau)|^2 d\tau \quad (5)$$

238 In analogy to the signal analysis procedures used in [18], [21], the Wavelet Transform is
 239 preferred to STFT for time-frequency analysis because it allows to dynamically increase the
 240 frequency resolution at lower frequency values, whereas to increase time resolution at higher
 241 frequency values (in STFT a the frequency resolution is fixed). Moreover, compared to classical
 242 Fourier spectra, the wavelet-based procedure allows to obtain smoother spectra and well
 243 defined from the mathematical point of view. The results in the following section are reported
 244 in form of time-frequency energy maps $|W_x|^2$, (also called ‘‘scalograms’’) and wavelet spectra
 245 P_{W_x} . In the scalogram, the colour represents the energy value at the given time and frequency.
 246 The related spectrum will be used for the identification of the dominant frequencies. Given that
 247 $W_x(a, \tau)$ and $W_y(a, \tau)$ are, respectively, the complex wavelet transforms of two signals $x(t)$
 248 and $y(t)$, their wavelet cross-scalogram is defined, as:

$$\begin{aligned} W_{x-y}(a, \tau) &= W_x^*(a, \tau) W_y(a, \tau) = \\ &CoW_{xy}(a, \tau) - i QuadW_{xy}(a, \tau) \end{aligned} \quad (6)$$

249 It can be shown that the real part of the cross-scalogram, or co-scalogram CoW_{xy} , gives the
 250 instantaneous contribution at each frequency scale to the correlation between the two signals.
 251 The instantaneous angle of phase, $\theta_{x-y}(a, \tau)$, can be expressed as:

$$\theta_{x-y}(a, \tau) = tg^{-1} \frac{QuadW_{xy}(a, \tau)}{CoW_{xy}(a, \tau)} \quad (7)$$

252 This formulation may be used to analyse the correlation between the fluctuations at a given
 253 frequency present in two simultaneously-acquired signals. When $\theta_{x-y} = 0 \pm k360 \text{ deg}$ the
 254 analyzed signals are perfectly in phase, whereas $\theta_{x-y} = 180 \pm k360 \text{ deg}$ indicates perfect
 255 phase opposition.

256 3 Results and discussion

257 The Wavelet analysis is performed on the pressure (P_f), fluid temperature (T_f) and wall
 258 temperature (T_w) signals. The analysis is performed on a frequency range from 0.01 to 4 Hz.
 259 For the processing of the signals, an in-house-developed wavelet tool has been used (see e.g.

260 [18], [21]), by imposing the following parameters show in Table 3, where number of samples
 261 is $N_{WT} = 2^n = f_{ac}t_{an}$.

Table 3. Wavelet parameters.

a	401		
ω_0	2π		
f_{\min} [Hz]	0.01		
f_{\max} [Hz]	4		
	Pressure	Temperature	g-level
f_{ac} [Hz]	200	50	5
n	12	10	7
N_{Wt}	4096	1024	128
t_{an} [s]	20.48	20.48	25.6

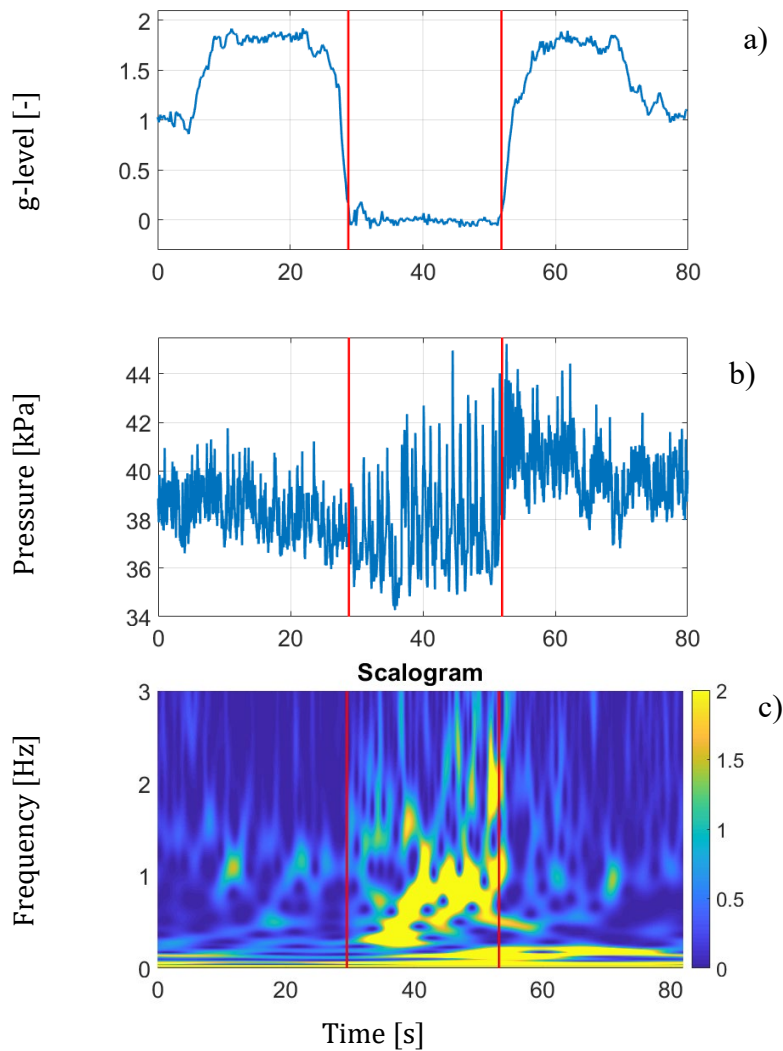
262 The Wavelet analysis is conducted through the following steps:

- 263 a) choice of the relevant time range (Microgravity period);
 264 b) definition of the dominant frequency;
 265 c) analysis of gravity acceleration as a possible disturbance;
 266 d) comparison of the Wavelet Scalogram and the Power Spectrum of the fluid pressure, fluid
 267 temperature and wall temperature signals;
 268 e) determination of the Wavelet Scalogram and the Power Spectrum of the fluid pressure signal
 269 at each heat input level;
 270 f) identification of the dominant frequency of the fluid pressure signal for each heat input level;
 271 g) comparison between the fluid pressure signal recorded at the evaporator and the signal
 272 recorded at the condenser through wavelet cross scalogram, wavelet local correlation
 273 coefficient and instantaneous angle of phase;
 274 h) identification of the mean instantaneous angle of phase at the dominant frequency between
 275 the fluid pressure signal recorded at the evaporator and at the condenser for each heat load
 276 input;
 277 i) identification of the mean time lag and then the mean velocity at the dominant frequency
 278 between the fluid pressure signal recorded at the evaporator and the signal recorded at the
 279 condenser for each heat load input.

280 **3.1 The microgravity period**

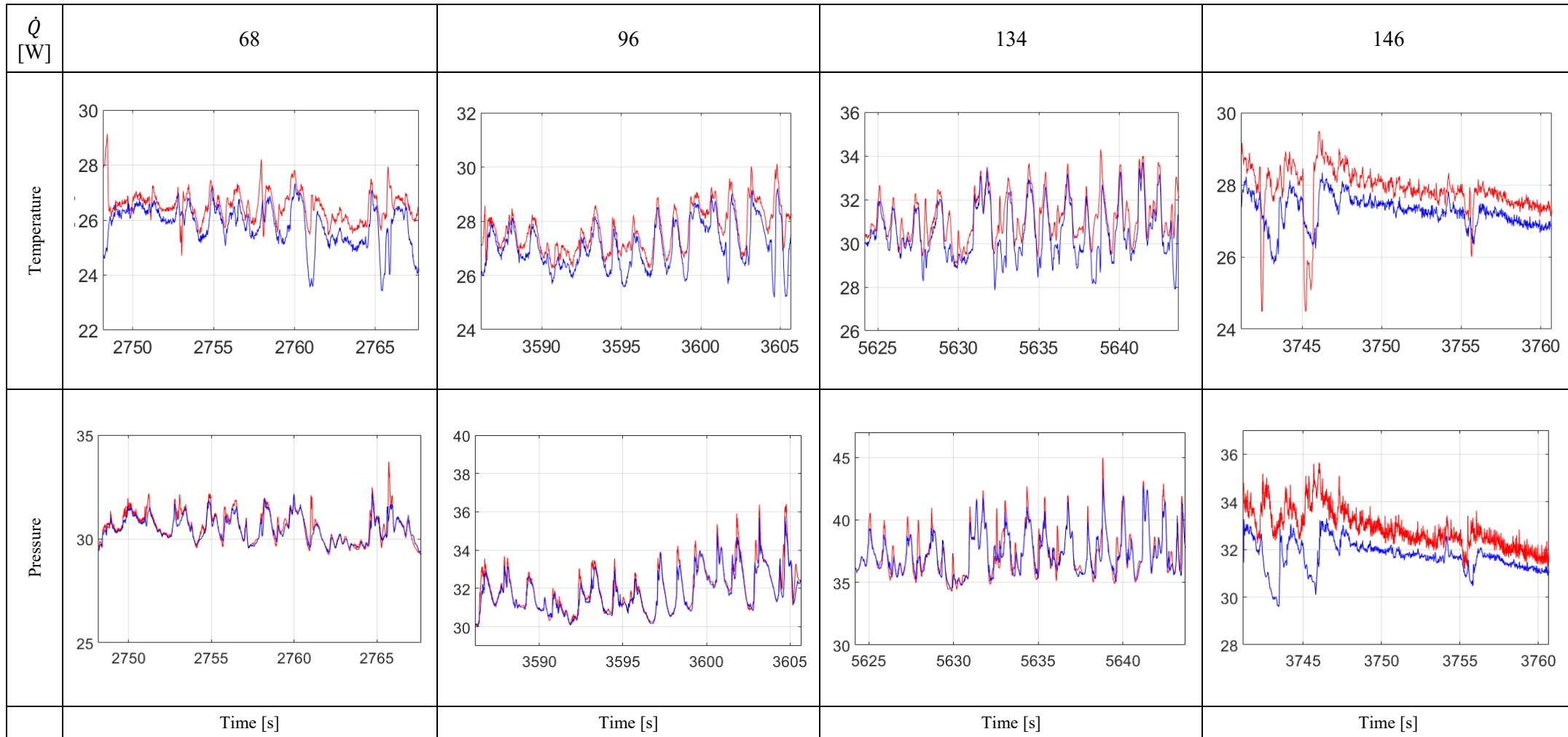
281 As thoroughly explained in [19], since the tube diameter is larger than the capillary limit on
 282 ground, this kind of Pulsating Heat Pipe has been actually proved to operate as a series of Loop
 283 Thermosyphons when gravity is present, while it switches to the Pulsating Heat Pipe mode
 284 (oscillating slug flow) when the gravity field is absent, i.e. during the microgravity period of a
 285 parabolic flight. In this framework, the time frequency analysis of the fluid pressure signal gives
 286 a further proof of such a change in the operating mode. Figure 4a shows the temporal trend of

287 the gravity field during a parabolic manoeuvre consisting of three periods: the first hyper-
 288 gravity (20 s) during the aircraft ascent, the microgravity period at the top of the parabola (20s)
 289 and the second hyper-gravity period (20 s) during the aircraft recovery. Figure 4b shows the
 290 temporal trend of the fluid pressure signal close to the evaporator zone synchronized with the
 291 gravity trend, where the red lines highlight the microgravity period. The sudden change in the
 292 fluid pressure fluctuation is clearly visible in terms of amplitude during the microgravity period
 293 but it is only the wavelet processing of the fluid pressure signal that reveals, by means of the
 294 scalogram (Figure 4c), the existence of energy peaks, apparently at a given frequency only
 295 during the microgravity period. For these reasons the WT analysis have been performed for all
 296 the heat input levels only during the microgravity duration. **The different temporal trend of fluid
 297 temperature and fluid pressure at the evaporator zone (red line) and at the condenser zone (blue
 298 line), for each load input is shown in Figure 5.**



330 Figure 4: a) g-level and b) fluid pressure temporal trend during a typical flight parabola, c) Wavelet
 331 scalogram.

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335 Figure 5: temporal trend of the fluid pressure and fluid temperature at the evaporator zone (red line) and at the condenser zone (blue line), for each heat
 336 load input.

337 **3.2 Definition of dominant frequency**

338 Given the wavelet power spectrum $P_{Wt}(f)$, in the domain $\text{Dom}(P_{Wt})$, the dominant
 339 frequency (f_D) shall be defined as the frequency at which the absolute maximum of the
 340 function is $P_{Wt}(f)$ occur. Thus, from an analytical point of view:

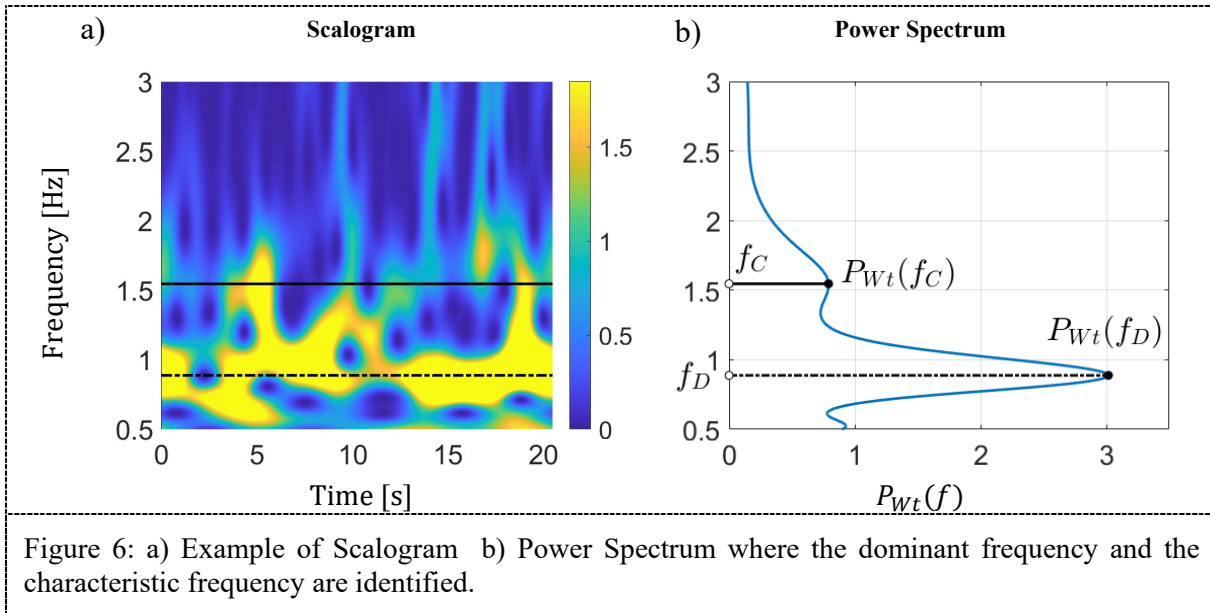
$$\forall f \in \text{Dom}(P_{Wt}) \quad P_{Wt}(f) < P_{Wt}(f_D) \quad (8)$$

341 Since the wavelet power spectrum $P_{Wt}(f)$ may present more than one peak, relative-maximum
 342 frequencies, also called characteristic frequencies, are also defined. The relative-maximum
 343 frequencies f_C exist if $\exists \text{SubDom}(P_{Wt})$ with $f_D \notin \text{SubDom}(P_{Wt})$ such that:

$$\forall f \in \text{SubDom}(P_{Wt}) \quad P_{Wt}(f) \leq P_{Wt}(f_C) \quad (9)$$

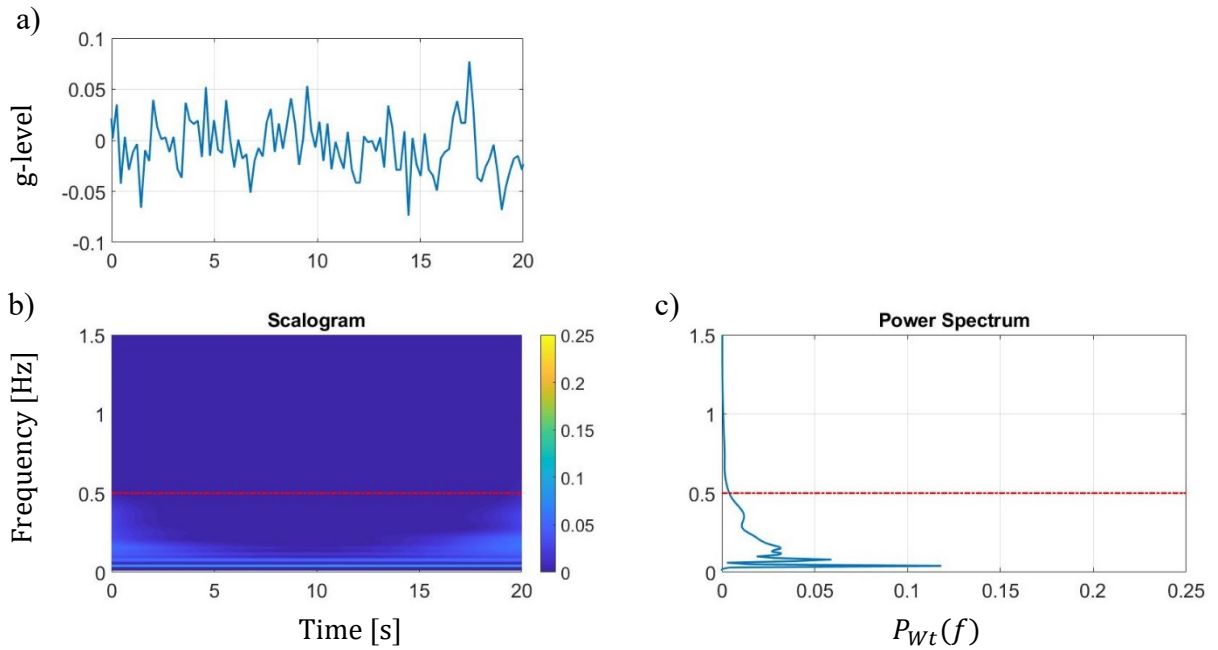
344 The dominant frequency (f_D) will be indicated in the Scalogram and in the Power Spectrum
 345 graphs as a black dashed line instead the characteristic, or secondary, frequency (f_C) as a black
 346 solid line (Figure 6b). Once defined the dominant and the characteristic frequencies from the
 347 wavelet spectrum, it is possible to evaluate also their temporal continuity from the scalogram.
 348 Notice that such a rigorous procedure allows to compare the results of different authors
 349 according to a unique analytical definition.

350



351 **3.3 Noise**

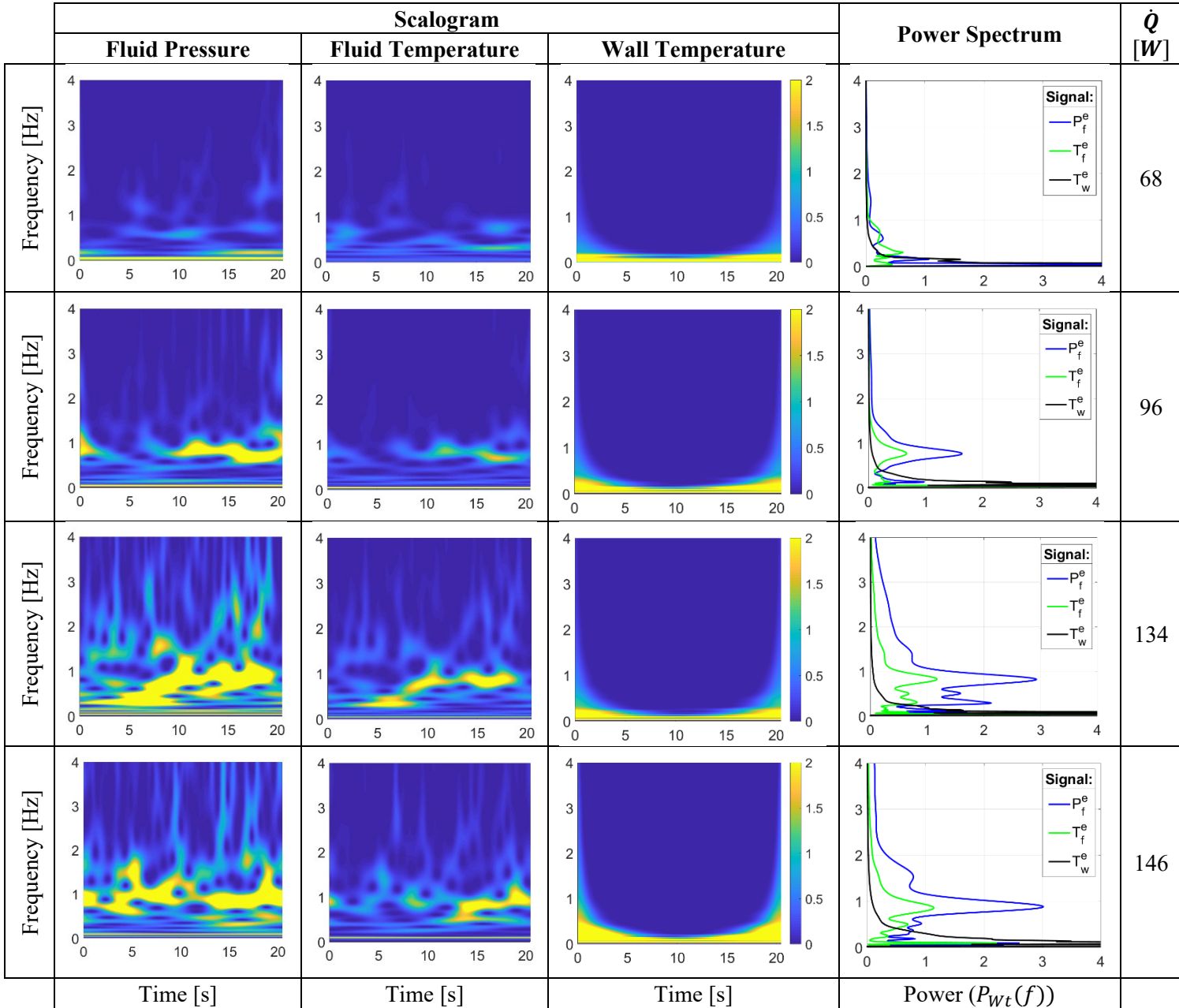
352 The gravitational acceleration g signal is analysed too by means of the WT method to ensure
 353 that the aircraft g -gitter does not add spurious frequencies to the pressure or temperature signals.
 354 In this case, the acquisition frequency is $f_{ac}^g = 50$ Hz, $N_{Wt}^g = 128$ is chosen for the analysis
 355 time $t_{an}^g = 25.6$ s, and with the g -level range $-0.1 \leq g \leq 0.1$ (Figure 6a). In the Wavelet
 356 Scalogram and in the Power Spectrum, there are not dominant or characteristic frequencies, as
 357 show in Figure 6 (b-c). Although the magnitude of the energy contribution of the g signal is
 358 almost negligible, the frequency limit $f \geq 0.5$ Hz (red hatch) ensures a noise-free analysis.



359 Figure 7: a) g-level temporal trend from the accelerometer; b) Wavelet Scalogram and c) Power
 360 Spectrum.

361 3.4 Signal choice

362 This section compares the results of WT analyses obtained using different signals, fluid pressure
 363 (P_f^e), fluid temperature (T_f^e) and wall temperature (T_w^e) signals recorded in the evaporator zone
 364 during the same time interval. For the pressure signal, the acquisition frequency is $f_{Ac}^P = 200$ Hz
 365 and $N_{Wt}^P = 4096$ is chosen for an analysis time $t_{An}^P = 20.48$ s. Instead, for the temperature
 366 signal, the acquisition frequency is $f_{Ac}^T = 50$ Hz and $N_{Wt}^T = 1024$ is chosen for an analysis
 367 time $t_{An}^T = 20.48$ s. The Scalograms for each signal are shown in Figure 8 and the Power
 368 Spectrum of all three signals in a unique plot (P_f^e , T_f^e , T_w^e respectively blue, green, black line).
 369 The fluid temperature signal appears to have the same dominant frequency as the pressure
 370 signal, with a lower energy content. The wall temperature signal, instead, does not show
 371 appreciable results in any frequency range, also according to Monroe et al. [13], and this is due
 372 to the thermal impedance of the tube wall: the PHP envelope and its thermal inertia act as a
 373 low-pass filter RC circuit on the signal. For these reasons, this work considers only the pressure
 374 signal at the evaporator P_f^e and condenser P_f^c .



376 Figure 8: Wavelet Scalogram and Power Spectrum of the fluid pressure signal (P_f^e blue line), the fluid
 377 temperature (T_f^e green line) and the wall temperature (T_w^e black line), for each heat load input.

378 **3.5 Pressure signal wavelet analysis**

379 The Wavelet Scalogram and Power Spectrum of a pressure signal at the evaporator section for
 380 each heat load level are represented in Figure 9, respectively. Also, the pressure signal at the
 381 condenser section is analysed, and the results are listed in Table 4. From a physical point of
 382 view, the two signals can be shifted or damped in terms of oscillation amplitude (Section 3.6),
 383 but the frequency is the same. The dominant frequency is identified by the black dashed line
 384 and by the value as shown in Section 3.2. The dominant frequency falls in the range 0.6 - 0.9
 385 Hz. Since each heat input level has been tested at least four times to guarantee repeatability, all

386 the results are analysed, compared and summarized in Figure 10 resuming the dominant
 387 frequency (y-axis) occurring at each power input (x-axis). A linear interpolation of all the data
 388 (dashed line in figure 10) shows that the value of the dominant frequency increases with
 389 increasing heat load input.

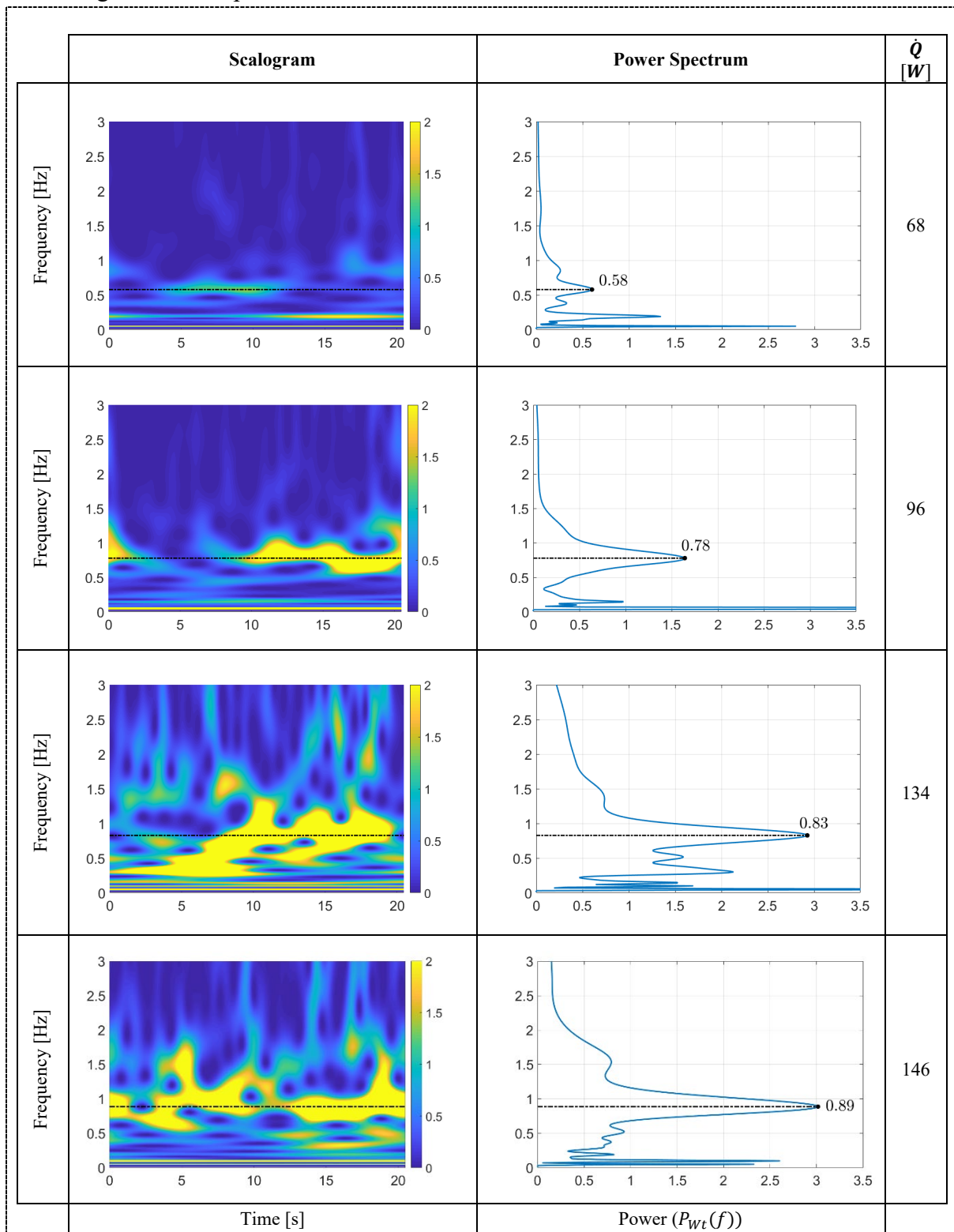


Figure 9: Wavelet Scalogram and Power Spectra of the fluid pressure signal (P_f^e) at each heat load input

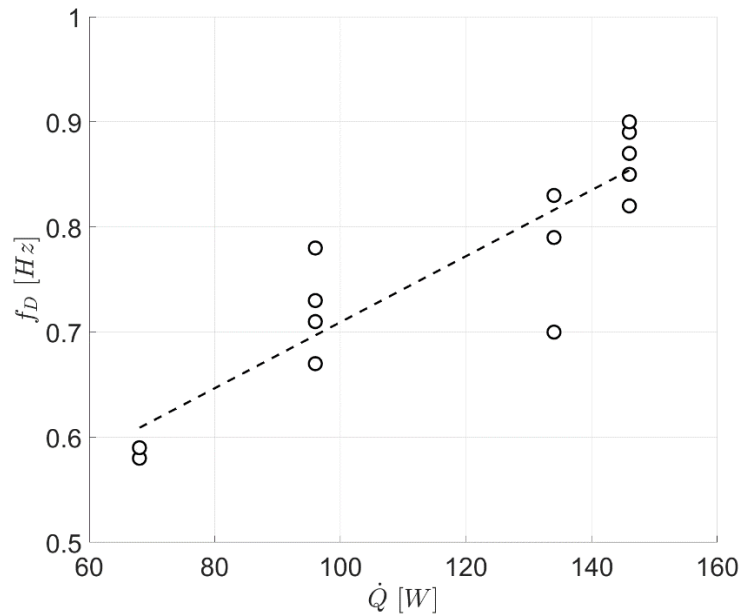


Figure 10: Dominant frequency trend with increasing heat load input of the fluid pressure signal.

390 3.6 Cross correlation and fluid mean velocity

391 As shown in Section 2.3, the co-scalogram of the cross-wavelet transform identifies the
 392 frequencies where energy peaks are present both for the evaporator (P_f^e) and the condenser (P_f^c)
 393 (Figure 11 a). Their contribution to the correlation is derived from the Instantaneous Angle of
 394 Phase, θ_{e-c} (Figure 11 b). As often happens, since trends are not clearly detectable, θ_{e-c} plots
 395 have been filtered by using the cross scalogram as mask (Figure 11 c).

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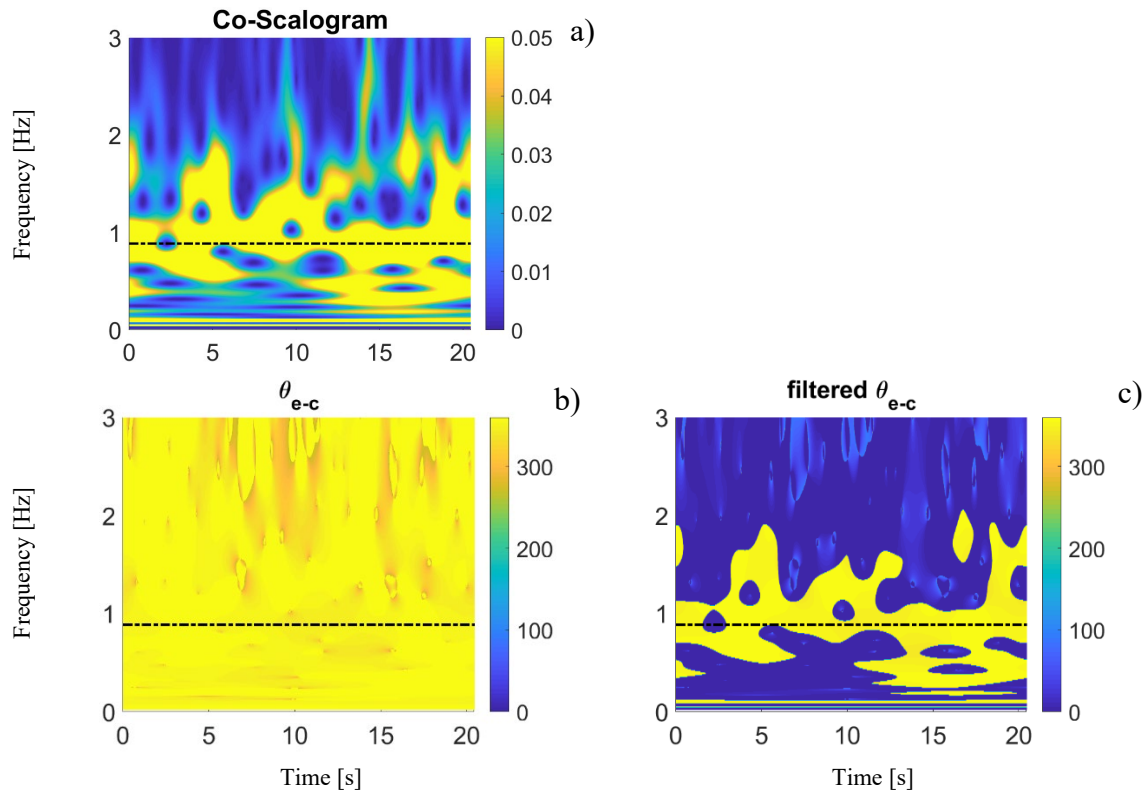


Figure 11: comparison between the fluid pressure signal at the evaporator (P_f^e) and the fluid pressure signal at the condenser (P_f^c). Wavelet Cross Scalogram (a). Instantaneous angle of phase θ_{e-c} (b) and related filtered (c).

407 Later, these filtered plots are integrated over time, obtaining the mean filtered $\bar{\theta}_{e-c}|_{f_D}$ plot as
 408 shown in the second column of Figure 12. The plots in the third column display the residual
 409 points percentage (r.p. [%]), i.e. the fraction of meaningful data after the filtering procedure.
 410 The r.p. percentage obtained at 68 W is below the 50%, meaning that data are not reliable after
 411 the filtering procedure. All the other power levels show r.p. percentages above 70%, meaning
 412 that the most of data fall close to the dominant frequency. The mean filtered $\bar{\theta}_{e-c}|_{f_D}$ is between
 413 350 deg and 355 deg at high heat load inputs and 311 deg at low heat load input.

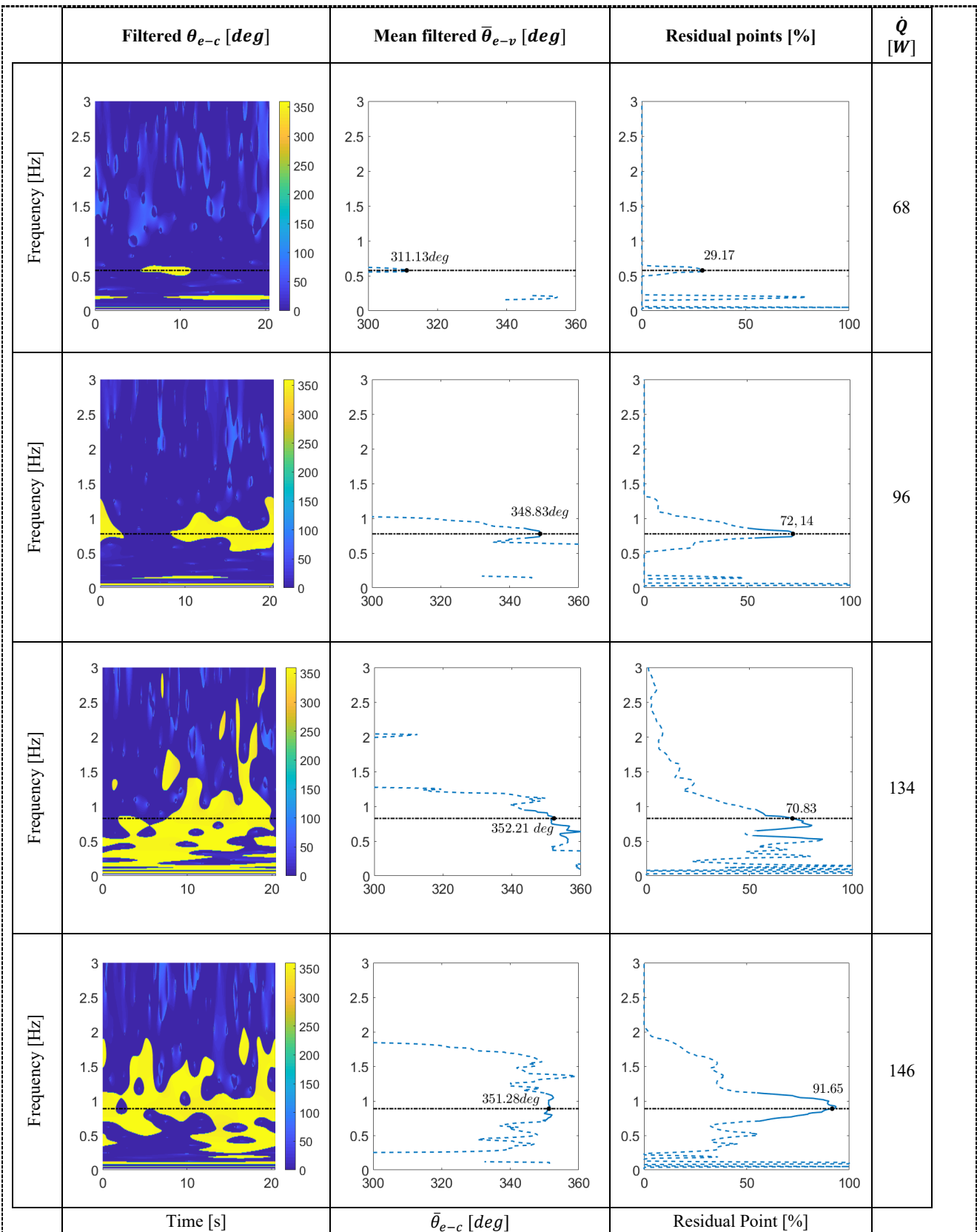


Figure 12: Filtered instantaneous angle of phase θ_{e-c} , mean filtered instantaneous angle of phase $\bar{\theta}_{e-c}$ and residual point for each heat load input.

414 Moreover, assuming the two signals have the same dominating frequency f , the time lag
 415 between two signals can be evaluated. In particular, the mean time lag $\bar{t}_{e-c}|_f$ between the two
 416 signals $P^e(t)$ and $P^c(t)$ at a given frequency f is:

$$\bar{t}_{e-c}|_{f_D} = \frac{\bar{\theta}_{e-c}|_{f_D}}{360f_D} \quad (10)$$

417 where $\bar{\theta}_{e-c}|_f(a)$ is the mean angle over time at the given frequency f (one value for each given
 418 a). Values of $\bar{t}_{e-c}|_{f_D}$ fall in the range 1.5 – 1.1 s. Assuming a Eulerian frame, the mean phase
 419 shift between the two pressure signals can be interpreted as the mean time needed by a batch of
 420 fluid to travel from the evaporator to the condenser. So, in first approximation, the mean time
 421 shift between the two pressure signals can be related to the fluid mean velocity of the fluid batch
 422 moving between the evaporator and the condenser as follows:

$$\bar{v}_{e-c}|_{f_D} = \frac{L}{\bar{t}_{e-c}|_{f_D}} \quad (11)$$

423 where $L = L_{PHP} = 0.14 \text{ m}$ is the distance between the evaporator and condenser measurement
 424 points as shown in figure 1.

Table 4. Resume of the wavelet analysis results.

DAY	p	\dot{Q} [W]	f_D^e [Hz]	f_D^c [Hz]	$r.p. _{f_D}$ [%]	$\bar{\theta}_{e-c} _{f_D}$ [deg]	$\bar{t}_{e-c} _{f_D}$ [s]	$\bar{v}_{e-c} _{f_D} = \bar{v}_{WT}$ [m/s]	\bar{v}_{IR} [m/s]
I	8	68	0.58	0.58	29.17	311.13	1.49	0.09	0.08
I	24	68	0.59	0.58	25.88	338.91	1.60	0.09	0.06
I	11	96	0.78	0.78	72.14	348.83	1.25	0.11	0.12
I	12	96	0.73	0.73	57.64	355.49	1.36	0.10	0.08
I	14	96	0.71	0.71	80.52	353.55	1.39	0.10	-
I	15	96	0.67	0.68	47.46	344.90	1.43	0.10	0.08
I	16	134	0.70	0.70	77.39	357.06	1.42	0.10	-
I	17	134	0.83	0.83	70.83	352.21	1.18	0.12	0.11
I	18	134	0.83	0.82	69.02	352.31	1.18	0.12	-
I	19	134	0.79	0.79	80.47	342.48	1.21	0.12	-
II	11	146	0.85	0.86	93.92	353.96	1.16	0.12	-
II	12	146	0.89	0.88	91.65	351.28	1.10	0.13	0.13
II	13	146	0.82	0.84	72.46	352.38	1.20	0.12	0.11
II	14	146	0.87	0.70	80.66	356.00	1.14	0.12	0.11
II	15	146	0.90	0.87	81.23	354.87	1.10	0.13	-

425 The mean velocity $\bar{v}_{e-c}|_{f_D}$ have been calculated for each parabola and falls in the range 0.09 -
 426 0.13 m/s. All the data collected are summarized in the Table 4. This order of magnitude is
 427 comparable with the velocity values obtained with infrared analyses described in subsection
 428 2.2, summarized in the last column of Table 4.

429 The velocity values achieved from the wavelet analysis ($\bar{v}_{e-c}|_{f_D} = \bar{v}_{WT}$) and from the infrared
 430 analysis (\bar{v}_{IR}) were also compared in the v/\dot{Q} plot (Figure 13).

431

432 Figure 13 shows that the value of the mean velocity increases with increasing heat load input.
 433 The repeatability of the results is indicated by the white/grey/black colours. For instance, as
 434 show in Table 4, at $\dot{Q} = 134 \text{ W}$, the results $\bar{v}_{e-c}|_{f_D} = 0.12 \text{ m/s}$ is repeated three times (black
 435 dots in Figure 11), while $\bar{v}_{e-c}|_{f_D} = 0.10 \text{ m/s}$ is repeated one time (white dot in Figure 13).
 436 The results shows that the velocity is not much sensitive to frequency variations of $\Delta f =$
 437 0.04 Hz ($0.79 < f_D < 0.83$), while it varies for $\Delta f = 0.13 \text{ Hz}$ ($0.70 < f_D < 0.83$), at heat
 438 load input given ($\dot{Q} = 134 \text{ W}$).
 439

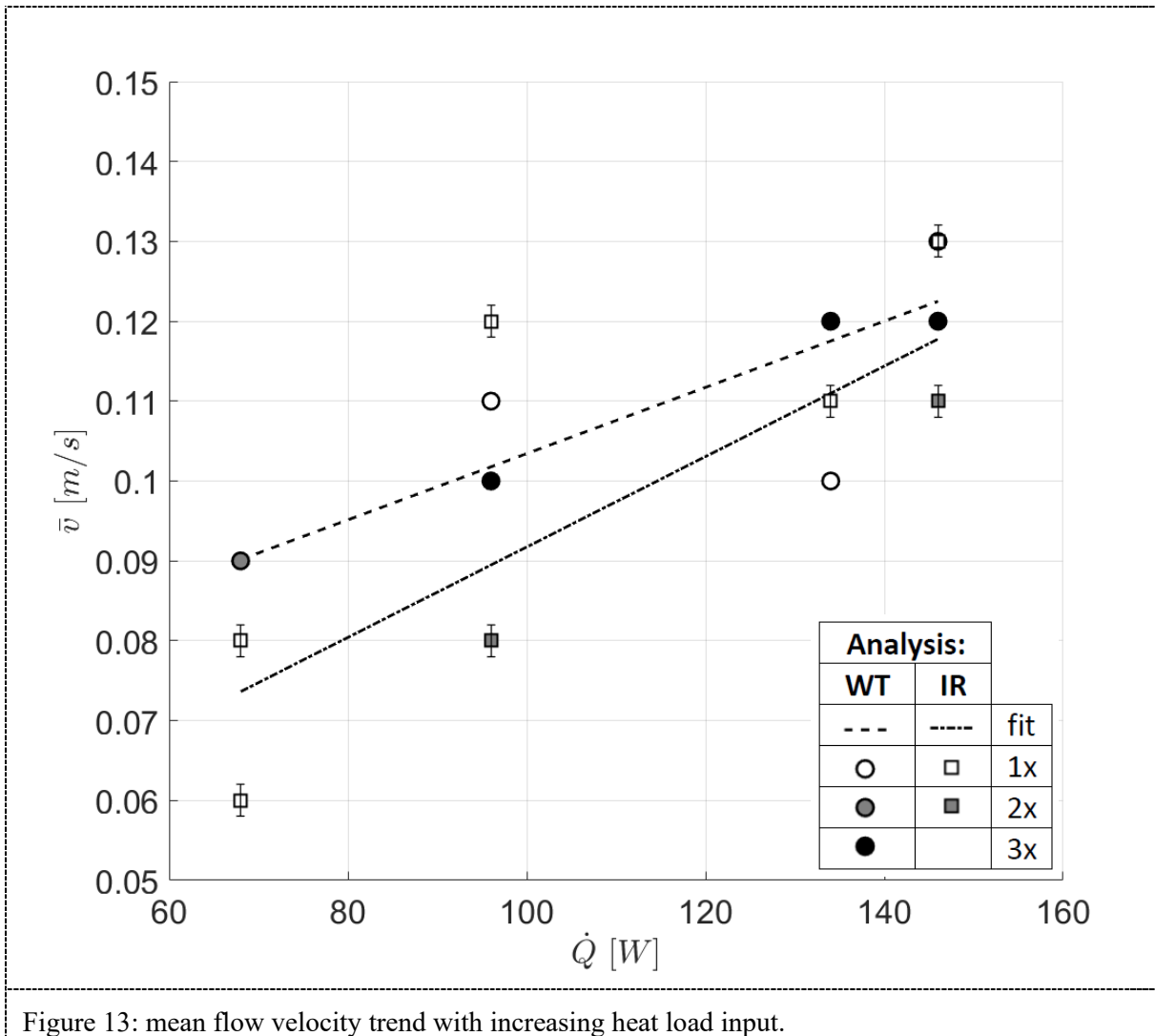


Figure 13: mean flow velocity trend with increasing heat load input.

440 Investigating the fluid motion by means of two pressure transducers combined with the time
 441 frequency analysis, constitute a novel and promising tool that allows to provide an order of
 442 magnitude of the fluid mean velocity without any visualization technique, thus simplifying the
 443 experimental hardware.

444 4 Conclusions

445 The fluid pressure signals at the evaporator and the condenser of a Pulsating Heat Pipe tested
 446 in microgravity conditions and with varying heat load input have been processed through a

447 wavelet analysis to provide a rigorous definition of dominant frequency in the fluid motion.
448 The investigation also allowed to relate the dominant frequencies to the heat input levels and,
449 finally to provide an esteem of the mean flow velocity through the cross correlation between
450 the two signals. In the light of literature review, the main outcomes of the present work are:

- 451 • the wall temperature signal of the hybrid PHP cannot be used for Wavelet analysis, due to
452 the thermal impedance of the tube wall while the pressure signal is the most suitable;
- 453 • in general, the dominant frequency should be defined as the frequency related to the
454 absolute maximum in the Power spectrum;
- 455 • the value of the dominant frequency of the pressure signals falls in the range 0.6-0.9 Hz
456 and increases with increasing heat load input;
- 457 • the cross correlation between the two pressure signals allow to evaluate the mean phase
458 shift, which itself can be related to the mean oscillating flow velocity. For the present case
459 it falls in the range 0.09 - 0.13 m/s and increases with increasing heat load input which is
460 physically consistent;
- 461 • the velocity values obtained from the wavelet analysis and from the infrared analysis were
462 in good agreement.

463 The possibility to investigate the average local fluid mean velocity only by means of pressure
464 signals processed with a wavelet tool, constitutes an important step forward the
465 characterization of the PHP behaviour and towards the validation of analytical and numerical
466 PHP models. Furthermore, this method may in some case replace the visualization inserts and
467 peripheral facilities (i.e. cameras). The complexity and cost of the measurements may be
468 effectively reduced, leading to greater overall reliability both in the scientific and industrial
469 field.

470

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